

AIRWORTHINESS DIRECTIVE COMPLIANCE RECORD - ENGINE STATUS (CLOSED,OPEN,REPEAT)											AIRCRAFT -----
ENGINE TSN : 50,767					ENGINE CSN : 33,998				AS OF DATE : 23-APR-19		
MAKE : CFM INTERNATIONAL					MODEL : CFM56-3C-1				SERIAL NO. : 856131		
									POSITION NO. : #2		
TYPE	AD NUMBER	AD EFFECTIVE DATE	SERVICE BULLETIN REFERENCE	SUBJECT	MONTHS/ DAYS/CYCLE OR HOURS OF COMPLIANCE (INTERVAL)	ACCOMPLISHED DATE/CYCLE HOURS OF COMPLIANCE / EFFECTIVE DATE	METHOD OF COMPLIANCE	STATUS	NEXT COMPLIANCE DUE DATE/HOURS/CYCL ES	COMPONENT REMARKS	AD REMARKS
EASA AD	2012-0209		SB CFM 72-1129 & CFM SB 72-0564 & SB 72-0879	ENGINE - ACCESORY GEARBOX (AGB) HAND-CRANKING- CRANKING PAD - MODN.	1)CAT 2A - AGB PN 335-300- 112-0 TO BE INSTALLED AT THE FIRST ENG. QUALIFYING SHOP VISIT AFTER 1 MAR 2012.(CFM56 3C1) 2) CFM56-7B-REFER COMPT REMARKS.		M12008 (CFM6-3C1) & M12007 (CFM56-7B)	O	0 0	OPEN (AT NEXT SHOP VISIT)	CFM56-3C1 & CFM56-7B
FAA AD	84-26-03		CFMI SB 73-021R1 CFMI SB 73-024R1 CFMI SB 73-034R1	TO PREVENT FLAMEOUTS IN ENGINE EQUIPPED WITH FUEL INJECTORS WITHOUT HEAT SHIELD.				N/ A	0 0	N/A TO CFM56-3C1 ENGINES.	APPLICABLE TO ALL MODEL CFM56-2 ENGINES INCORPORATING FUEL INJECTOR P/NO.9984M90G20.
FAA AD	86-08-05		CFM SB 72-205R5	RADIAL DRIVE SHAFT DISENGAGE.	-	-		N/ A	0 0	N/A PER P/NO. APPLICABLE TO CFM56-3/-3B SERIES	N/A TO P/N 335-300-012 AND ENGINE MODEL
FAA AD	89-13-51		CFM SB 73-473R2	REPLACE STAGE 1 FAN DISK AND BLADE.	-	-		S	0 0	N/A DUE SUPERSEDED BY FAA AD 96-25-11.	SUPERSEDED BY FAA AD 96- 25-11.
FAA AD	89-17-04		CFM SB 72-445 CFM SB 72-578	TO PREVENT FAILURE OF NO. 3B BEARING.	-	-		S	0 0	THIS AD SUPERSEDED BY FAA AD 89-23-06 REFERS.	SUPERSEDED BY FAA AD 89- 23-06.
FAA AD	89-23-06 R1	1/28/1998	CFM SB 73-530R3	NO. 3 BEARING IMPROVEMENTS.	-			S	0 0	N/A TO NO. 3 BEARING INSTALLED, P/N 1461M16P0P	SUPERSEDES FAA AD 89-17- 04.
FAA AD	90-20-13	#####	CFM SB 72-543R4 CFM SB 72-494R4	TO PREVENT FAN BLADE FAILURE, INFLIGHT SHUTDOWN AND SEVERE ENGINE DAMAGE.	-	-		S	0 0	N/A - APPLICABLE TO S/NOS 725205 TO 725219, 725242 TO 725243.	SUPERSEDES SB 72-494 R3. FAN BLADES P/NO. NOT AFFECTED. DAMPERS PREVIOUSLY INSTALLED PER SB 72-A916
FAA AD	91-02-10	2/11/1991	CFM SB 72-450R3 CFM SB 72-462R2	INTRODUCTION OF NEW SPLITTER FAIRING.	-	-		N/ A	0 0	N/A	N/A POST SB 72-450R3 : N/A POST SB 72-462R2 :
FAA AD	93-05-05	4/29/1993	CFM SB 72-580R6	ENGINE MODIFICATION FOR INCREASED CAPABILITY IN HAIL, RAIN AND SLEET.	-			C	0 0		DONE IN PRODUCTION. CFM56-3C SERIES.
FAA AD	95-01-05	1/20/1995	-	TO PREVENT DAMAGE TO THSE ENGINES DUE TO ICE INGESTION	-			N/ A	0 0	N/A	NOT APPLICABLE TO CFM56 SERIES ENGINE
FAA AD	96-18-16	12/9/1996	CFM SB 72-695	TO PREVENT A LOW CYCLE FATGUE (LCF) FAILURE OF THE LOW PRESSURE TURBINE ROTOR (LPT) STUB SHAFTS AND CONICAL SUPPORTS.	-	-		N/ A	0 0	N/A DUE LPT CONICAL SUPPORTS AND LPT STUB SHAFT S/NOS ARE NOT AFFECTED AS LISTED IN SB LISTING.	N/A DUE LPT CONICAL SUPPORTS AND LPT STUB SHAFT S/NOS. ARE NOT AFFECTED AS LISTED IN SB LISTING.
FAA AD	96-25-11	1/29/1997	BSB 71-1203R10 CFM SB 72-543R4	INTRODUCTION OF FAN BLADES WITH 37 DEGREES MID SPAN SHROUDS.	-			N/ A	0 0	N/A - APPLICABLE TO S/NO. 724581 TO 724698, 725116 TO 725685.	NA TO ENGINE CONFIGURATION. FAN BLADE POST SB 72-0543. P/N NOT INSTALLED
FAA AD	97-05-13		CFM SB 80-003	TO PREVENT AN AIR TURBINE ENGINE STARTER FAILURE.	-	-		N/ A	0 0	N/A TO CFM56-3 SERIES.	APPLICABLE TO CFM56-5 SERIES ENGINE ONLY.
FAA AD	97-06-01		CFM SB 72-440 CFM SB 72-064 CFM SB 72-229	TO PREVENT A FAILURE OF THE STAGE 1 DISK OF THE HPCR STAGE 1-2 SPOOL.	-	-		N/ A	0 0	N/A TO CFM56-3 SERIES.	APPLICABLE TO CFM56-5, - 5B AND -5C SERIES ENGINE ONLY.

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FAA AD	97-08-01	6/23/1997	CFM1.TP.SM.5	TO PREVENT A LOW CYCLES FATIGUE (LCF) FAILURE OF THE FAN DISK.	-		-	C	0 0	CFM INTERNATIONAL (CFMI) CFM56-3, -3B, AND -3C SERIES TURBOFAN ENGINES WITH FAN DISKS, PART NUMBER (P/N) 335-014-509-0 OR 335-014-511-0, INSTALLED	CH 05-11-01 HAS BEEN UPDATED BY CFMI. NEW LLP STATUS (UKR AD STATUS)
FAA AD	97-09-02		-	TO PREVENT A LOW CYCLES FATIGUE (LCF) FAILURE OF THE HPTR FRONT SHAFT, HPTR FRONT AIR SEAL, HPTR DISK, BOOSTER SPOOL AND LPTR STAGE 3 DISK.	-		-	N/A	0 0	N/A TO CFM56-3C ENGINES.	N/A DUE APPLICABLE TO CFM56-5C2/G, -5C3/G AND -5C4 SERIES ENGINE.
FAA AD	97-25-51		CFM ASB 72-A861R2 CFM ASB 72-863	TO PREVENT INFLIGHT ENGINE SHUTDOWNS DUE TO AN AGE STARTER GEARSHAFT OF TGB INPUT BEVEL GEAR FAILURE.	AD SUPERSEDED BY AD 98-10-11.		-	N/A	0 0	N/A TO THIS ENGINE S/NO.	N/A DUE APPLICABLE TO CFM56-3, -3B AND -3C SERIES ENGINES HAVING S/NOS. 858619 TO 662, 858624, 858627 TO 635/638/640/641/644/647/856625, 858630 TO 639, 858641 TO 643, 858646 TO 651, 858654 TO 657, 858660 TO 662, 858665 TO 668/673/687/688 AND 701.
FAA AD	98-02-04		-	TO PREVENT A LOW CYCLES FATIGUE (LCF) FAILURE OF THE LPT CASE.	-		-	N/A	0 0	N/A TO THIS ENGINE SERIES.	N/A DUE APPLICABLE TO CFM56-5B1/2P, -5B2/2P AND -5B4/2P SERIES ENGINE.
FAA AD	98-07-02	3/30/1998	CFM SB 72-856 CFM SB 72-855R1-SUPERSEDED BY SB 72-0920.	TO PREVENT RUBS BETWEEN THE OUTER CONE OF THE NO. 3 BEARING REAR STATIONARY.	CAT.4A - PARA 'B' & C ITEM 2 AT NEXT SHOP VISIT -ESN AFFECTED - CAT.2A - PARA. 'C' ITEM 1 WITHIN 15 DAYS AFTER 30/03/98. (AFFECTED ESN : 856692/709/713/799/673/691/694/696/697/746/780/857669/ 685/686/704 AND 859115).			N/A	0 0	N/A PER NO. 3 BEARING AFT STATIONARY A/O SEAL P/N 1663M91G04 NOT EFFECTED	N/A PER NO. 3 BEARING AFT STATIONARY A/O SEAL P/N 1663M91G04 INSTALLED
FAA AD	98-10-11	6/3/1998	CFM ASB 72-A861R3 CFM SB 72-863R1 CFM SB 72-865 CFM SB 72-867 CFM SB 72-873R1	TO PREVENT INFLIGHT ENGINE SHUTDOWN DUE TO AN ACCESSORY GEARBOX STARTER GEARSHAFT, TRANSFER GEARBOX INPUT BEVEL GEAR, TGB OUTPUT BEVEL GEAR, AGB GEARSHAFT CLUSTER SPUR ASSY OR AGB INTERMEDIATE GEAR ASSY FAILURE.	-		-	N/A	0 0	N/A DUE ESN NOT IN THE AFFECTED RANGE.	N/A DUE ESN NOT IN THE AFFECTED RANGE.THIS AD SUPERSEDES AD 97-25-51.

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FAA AD	98-12-32		CFM SB 72-843R1	TO PREVENT THE POTENTIAL FOR AN UNCONTAINED FAILURE OF THE HPTR DISK.	-		-	N/A	0 0	N/A PER P/N 1475M29P02, S/N GWN0209C INSTALLED	N/A PER P/N 1475M29P02, S/N GWN0209C INSTALLED	
FAA AD	98-14-51		CFM SB 72-130 CFM SB 72-132	TO PREVENT A POSSIBLE DUAL INFLIGHT ENGINE SHUTDOWN EVENT WHICH COULD RESULT IN A FORCE LANDING AND LOSS OF THE AIRCRAFT.	-		-	N/A	0 0	N/A TO CFM56-3C SERIES.	APPLICABLE TO CFM56-7B SERIES ENGINE.	
FAA AD	98-18-51		-	TO PREVENT AN UNCOMMANDED ENGINE ACCELERATION EVENT OR INFLIGHT ENGINE SHUTDOWN.	-		-	N/A	0 0	N/A TO CFM56-3C ENGINES.	N/A DUE APPLICABLE TO CFM56-7B SERIES ENGINE.	
FAA AD	98-19-10	9/28/1998	CFM SB 72-877R1	ENGINE-ACCESSORY GEARBOX (72-63-00) STARTER	BEFORE FEB 1999.			N/A	0 0	N/A DUE ESN NOT AFFECTED.	N/A DUE ESN NOT AFFECTED.	
FAA AD	98-19-20		CFMI SB 73-016R2	TO PREVENT FAILURE OF THE HYDROMECHANICAL UNIT OVERSPEED GOVERNOR SPOOL VALVA SHAFT AND SUBSEQUENT ENGINE FLAMEOUT.	WITHIN 300 HRS FROM LAST INSP. PER SB.		-	N/A	0 0	NOT APPLICABLE TO CFM56-3 SERIES.	FOR CFM56-7B AND -7B2 ENGINES ONLY.	
FAA AD	98-21-23		ALL OPERATOR'S WIRE 98/CFM56/312R1 CFM56-7A ASB NO.73-A024	TO PREVENT AN UNCOMMANDED ENGINE ACCELERATION EVENT OR INFLT ENGINE SHUTDOWN.	WITHIN 20 CYC FROM 02/11/98.		-	N/A	0 0	NOT APPLICABLE TO CFM56-3 SERIES.	ALL CFM56-7B SERIES ENGINE. THIS AD SUPERSEDES AD T98-18-51.	
FAA AD	99-08-16		-	REQUIRED ENHANCE INSPN OF SELECTED CRITICAL LIFE LIMITED PARTS.	INITIAL : AT NEXT ENGINE SHOP VISIT OR PIECE PARTS OPPORTUNITY. RPT : EVERY ENGINE SHOP VISIT OR PIECE PARTS OPPORTUNITY.			S	0 0	THIS AD SUPERSEDED BY FAA AD 2000-12-01.	APPLICABLE TO CFM56-2, -2A, -2B, -3, -3B AND -3C SERIES. THIS AD IS SUPERSEDED BY FAA AD 2000-12-01.	
FAA AD	2000-05-22	5/2/2000	CFM SB 72-922	ECI INSPN FOR CRACKS IN BOLT HOLES OF HPT FRONT	-		-	N/A	0 0	N/A P/N NOT INSTALLED	N/A - INSTALLED P/N 1282M72P05, NOT	
FAA AD	2000-12-01		-	TO PREVENT CRITICAL LIFE-LIMITED ROTATING ENGINE PART FAILURE.	INITIAL : AT NEXT ENGINE SHOP VISIT OR PIECE PARTS OPPORTUNITY. RPT : EVERY ENGINE SHOP VISIT OR PIECE PARTS OPPORTUNITY.			S	0 0	SUPERSEDED	THIS AD SUPERSEDES FAA AD 99-18-16. THIS AD IS SUPERSEDED BY AD 2002-13-03	
FAA AD	2000-15-01		SB CFM 73-A129 SB AGT 708600-73-110	TO PREVENT FUEL LEAKAGE FROM BET. THE FUEL PUMP FILTER COVER AND GEAR HOUSING.	CAT 2A NEXT SHOP VISIT OF ENGINE, MODULE OR FUEL PUMP WHICHEVER OCCURS FIRST.			N/A	0 0	N/A	N/A INSTALLED P/N 708600-7 NOT EFFECTED	
FAA AD	2001-02-12	2/14/2001	CFM56 SB 75-0005	PREVENTION OF AIR LEAKAGE FROM INCORRECTLY TORQUED FITTINGS OF THE PS3 LINE.	WITHIN 25 DAYS FROM 14/02/2001.			N/A	0 0	N/A TO CFM56-3C1 ENGINES.	APPLICABLE TO CFM56-7B ENGINE FITTED TO 8737-600/700/800/900.	
FAA AD	2001-04-06	4/4/2001	CFMI SB 72-854R2 FAD 1997-298(B)R4	FAN DISK INSPECTION.	ONCE OFF BEFORE FAN DISK REACHES 22000 CYC OR 38500 HRS WHICHEVER COMES FIRST.	15-Sep-08		T	0 0	COMPLETED BY TAP (TAP AD STATUS FROM SHOP REPORT)	HAS BEEN COMPLIED WITH PER SB 72-854. APPL. TO CFM56-3, -3B AND -3C SERIES TURBOFAN ENGINES	
FAA AD	2001-09-17		CFMI ASB CFM56-5C & SB 73-A0106	TO PREVENT FUEL LEAKAGE ON THE HOT SECTION OR IN THE PRIMARY FIRE ZONE OF THE ENGINE.				N/A	0 0	N/A DUE APPLICABLE TO CFM 56-5C TURBOFAN ENGINES.	APPLICABLE TO CFMI CFM56-5C TURBOFAN ENGINES.	

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FAA AD	2001-11-05	6/11/2001	CFMI SB 72-A0965 CFMI SB 72-A0966	TO PREVENT NO.4 BEARING FAILURES WHICH COULD CAUSE AN ENGINE FAILURE.				N/A	0 0	N/A	N/A PER P/N 335-352-301-0 S/N: RR14092 INSTALLED	
FAA AD	2001-17-14		-	TO PREVENT AN LPT CONICAL SUPPORT FROM REMAINING IN SERVICE BEYOND ITS CERTIFIED CYCLIC LIFE LIMIT.	BEFORE ACCUMULATING 9350 CYC SINCE NEW.			N/A	0 0	APPLICABLE TO CFM56-5C4/1 SERIES ONLY.	APPLICABLE TO CFM56-5C4/1 SERIES ONLY. THIS AD SUPERSEDED BY 2007-24-04.	
FAA AD	2002-02-13	3/21/2002	HONEYWELL SB 3505582-80-1706	TO PREVENT UNCONTAINED FAILURE OF THE STARTER DUE TO LOSS OF OIL.	500 CYC FROM 21/03/2002.			N/A	0 0	N/A TO CFM56-3 SERIES ENGINES.	APPLICABLE TO CFM56-5 SERIES ONLY.	
FAA AD	2002-13-03	8/1/2002	ESM CFMI-TP.SM.5	TO PREVENT UNCONTAINED FAILURE OF CRITICAL LIFE LIMITED ROTATING ENGINE PARTS.	INITIAL AT NEXT ENGINE SHOP VISIT PIECE PART OPPORTUNITY REPEAT AT EVERY ENGINE SHOP VISIT PIECE PART OPPORTUNITY.	15/SEP/2008 28/SEP/2009		R	0 0	EVERY ENGINE SHOP VISIT PIECE PART OPPORTUNITY.	CFM56-2,-2A,-2B,-3,-3B,-3C,-5,-5B,-5C, AND -7B.THIS AD SUPERSEDES AD 2000-12-01.	
FAA AD	2002-16-18	9/18/2002	SB CFM56-7B-72-0241	2ND AND 3RD STAGE LOW PRESSURE TURBINE NOZZLE SEGMENTS - TO RETIRE FROM SERVICE OF THE AFFECTED SEGMENTS.	BEFORE ACCUMULATING 25000 CSN OR NOT LATER THAN 31/10/2008 W/EVER FIRST.			N/A	0 0	N/A TO CFM56-3C SERIES.	APPLICABLE TO CFM56-5B AND -7B SERIES ONLY.	
FAA AD	2003-02-04		SB 77-0020 SB 77-0008	TO PREVENT UNEXPECTED DETERIORATION OF CRITICAL ROTATING ENGINE PARTS DUE TO HIGHER THAN DESIRED ENGINE OPERATING EGT.	WITHIN 500 HRS FROM 27/02/03.		NIL	N/A	0 0	NOT APPLICABLE TO CFM56-3C SERIES.	APPLICABLE TO CFM56-5 AND -5B SERIES.	
FAA AD	2004-10-13	6/24/2004	SB 73-A129 SB 73-087 SB 73-0120R5(M04008) AGT SB 708600-73-102	TO PREVENT MAIN FUEL PUMP BEARING FAILURES RESULTING IN FUEL NOZZLE CLOGGING,LOW PRESSURE TURBINE(LPT) CASE BURN-THROUGH AND DAMAGE TO THE AIRPLANE.	NEXT ENGINE SHOP VISIT OR MAIN FUEL PUMP REPLACEMENT W/EVER EARLIER BUT NO LATER THAN 01/01/2007.	25-DEC-06	EO 002-06; JO E002/J-06	T	0 0	ENGINE FUEL MAIN PUMP P/N 708600-5 S/N 10324 INSTALLED (EO #002-06; JO E002/J-06)	ENGINE FUEL MAIN PUMP P/N 708600-5 S/N 10324 INSTALLED (EO #002-06; JO E002/J-06)	
FAA AD	2005/10/05		CFM SB 80-0018R1 CFM SB 80-0020R1 CFM SB 80-0011R1 CFM SB 80-0013R1	TO PREVENT ONCONTAINED FAILURES OF THE AIR TURBINE STARTERS WHICH COULD RESULT IN DAMAGE TO THE AIRPLANE.	NEXT SHOP VISIT BEFORE 31/12/2009.		NIL	N/A	0 0	NOT APPLICABLE TO CFM56-3C SERIES.	CFM56-5,-5B AND -5C. NOT APPLICABLE TO MH.	
FAA AD	2006-26-01	1/3/2007	73-010-01	TO PREVENT THE LOSS OF ENGINE THRUST THAT COULD RESULT IN LOSS OF CONTROL DURING TAKEOFF OR LANDING.	FILTER CHANGE AT 4000HRS.	25-DEC-06	REMOVE AFFECTED WESTERN AND PTI TECHNOLOGIES FUEL FILTERS	C	0 0	LISTED FILTER ELEMENTS ARE NOT USED I.A.W. MD001MO-2006	LISTED FILTER ELEMENTS ARE NOT USED I.A.W. MD001MO-2006	
FAA AD	2007-24-04	12/6/2007	-	TO PREVENT LPT CONICAL SUPPORTS FROM REMAINING IN SERVICE BEYOND THEIR CERTIFIED LIFE LIMIT.	DO NOT INSTALL LPT CONICAL SUPPORT PN 337-002-407-0 WITH 20000 OR MORE CSN.		NIL	N/A	0 0	NOT APPLICABLE TO THIS ENGINE SERIES.	CFM56-5C4/1 ENGINES ONLY.	

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FAA AD	2009/11/02	5/19/2009	NIL	TO PREVENT CRACKING OF THE HPC 4-9 SPOOL.	NIL		NIL	N/A	0 0	N/A-	N/A INSTALLED HPC 4-9 SPOOL P/N 1590M29G01, S/N GWN026CA NOT EFFECTED
FAA AD	2010/12/03	7/13/2010	SB 72-1067	FAILED FAN BLADE WITH SEVERE OUT-OF-LIMIT WEAR .	WITHIN 900 CYC FROM 13/07/2010.		NIL	N/A	0 0	N/A	N/A FAN MIDSPAN SHROUD FAN BLADES NOT EFFECTED
FAA AD	2012/02/03	3/15/2012	SB 72-0777 REV.01	TO PREVENT AN IFSD OF ONE OR MORE ENGINES FOLLOWING FOD OR A BIRD STRIKE.	CAT 7B		-	N/A	0 0	N/A	CFM 56-5B SERIES
FAA AD	2013/02/02	1/28/2013	-	TO PREVENT UNCONTAINED RELEASE OF MULTIPLE TURBINE BLADES, DAMAGE TO ENGINE, AND DAMAGE TO THE AIRPLANE.	CAT 7B		-	N/A	0 0	N/A	N/A INSTALLED HPT DISK P/N 1475M29P02, S/N GWN02EC1 NOT EFFECTED
FAA AD	2013/26/01	2/3/2014	-	TO PREVENT LOSS OF ENGINE OIL WHILE IN FLIGHT, WHICH COULD RESULT IN ENGINE FAILURE, LOSS OF THRUST CONTROL, AND DAMAGE TO THE AIRPLANE			-	N/A	0 0	N/A	N/A AGB P/N 335-300-205-0 NOT EFFECTED
FAA AD	2016-14-10	2/2/2013		TO PREVENT UNCONTAINED RELEASE OF MULTIPLE TURBINE BLADES, DAMAGE TO ENGINE, AND DAMAGE TO THE AIRPLANE.				N/A		N/A	N/A INSTALLED HPT DISK P/N 1475M29P02, S/N GWN0209C NOT EFFECTED
FAA AD	2017-14-08	8/18/2017		maintain the actuators ability to fully reach commanded position, and prevent LOTC and reduced control of the airplane.	Within 12 month		Inspect to determin HPC Stator Case Part Number	O	8//8/2018		